

Plates: normal load
differential formulation

*configuration variables: geometrical variables
 space primal complex*

*source variables: statical variables
 space dual complex*

Lagrange-Germain

$$\nabla^4 w(x, y) = \frac{q(x, y)}{D}$$

displacement method

1P

$$w$$

$$\begin{aligned} \chi_x &= -\partial_{xx} w \\ \chi_y &= -\partial_{yy} w \\ \chi_{xy} &= -\partial_{xy} w \end{aligned}$$

2L

$$\chi_x, \chi_y, \chi_{xy}$$

$$\begin{aligned} -\partial_y \chi_x - \partial_x \chi_{xy} &= \eta_x \\ -\partial_y \chi_{xy} - \partial_x \chi_y &= \eta_y \end{aligned}$$

1S

$$\eta_x, \eta_y$$

$\eta_x = 0$
 $\eta_y = 0$

$$\begin{Bmatrix} M_x \\ M_y \\ M_{xy} \end{Bmatrix} = \begin{bmatrix} D & \nu D & 0 \\ \nu D & D & 0 \\ 0 & 0 & (1-\nu)D \end{bmatrix} \begin{Bmatrix} \chi_x \\ \chi_y \\ \chi_{xy} \end{Bmatrix}$$

force method

1Ṽ

$$q$$

equilibrium

$$-\partial_{xx} M_x - \partial_{yy} M_y - 2\partial_{xy} M_{xy} = q$$

2S̃

$$M_x, M_y, M_{xy}$$

$$\begin{aligned} M_x &= \partial_y \cdot i_x \\ M_y &= -\partial_x \cdot i_y \\ M_{xy} &= \frac{1}{2} (\partial_y \cdot i_y - \partial_x \cdot i_x) \end{aligned}$$

1L̃

$$\cdot i_x, \cdot i_y$$

- w vertical deflection
- $\chi_x, \chi_y, \chi_{xy}$ flexional and twisting curvatures
- η_x, η_y incompatibility functions
- q normal load for unit surface
- M_x, M_y, M_{xy} bending and twisting moments
- $\cdot i_x, \cdot i_y$ stress functions
- D bending rigidity

$$D = \frac{Eh^3}{12(1-\nu^2)}$$